

AD 741905

ARL 72-0034

FEBRUARY 1972



Aerospace Research Laboratories

SOLAR RADIATION PRESSURE ON HIGH ALTITUDE SATELLITE

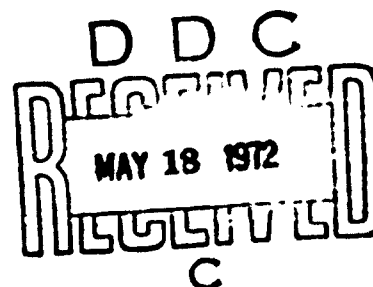
MOSHE CARMELI

GENERAL PHYSICS RESEARCH LABORATORY

Reproduced from
**NATIONAL TECHNICAL
INFORMATION SERVICE**
Springfield, Va. 22151

PROJECT NO. 7114

Approved for public release; distribution unlimited.



AIR FORCE SYSTEMS COMMAND

United States Air Force

16
R

NOTICES

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the United States Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto.

Agencies of the Department of Defense, qualified contractors and other government agencies may obtain copies from the

Defense Documentation Center
Cameron Station
Alexandria, Virginia 22314

This document has been released to the

CLEARINGHOUSE
U. S. Department of Commerce
Springfield, Virginia 22151

for sale to the public.

Copies of ARL Technical Documentary Reports should not be returned to Aerospace Research Laboratories unless return is required by security considerations, contractual obligations or notices on a specified document.

AIR FORCE: 8-5-72/200

ACQUISITION IN	WHITE CERTIFICATE	<input type="checkbox"/>	<input type="checkbox"/>
LIBRARY	SOFT CERTIFICATE		
REPRODUCED			
JUSTIFICATION			
BY			
RESTRICTION/AVAILABILITY CODES			
INT. MAIL			
DATE			

A

ARL 72-0034

SOLAR RADIATION PRESSURE ON HIGH ALTITUDE SATELLITE

MOSHE CARMELI

GENERAL PHYSICS RESEARCH LABORATORY

FEBRUARY 1972

PROJECT 7114

Approved for public release; distribution unlimited.

AEROSPACE RESEARCH LABORATORIES
AIR FORCE SYSTEMS COMMAND
UNITED STATES AIR FORCE
WRIGHT PATTERSON AIR FORCE BASE, OHIO

FOREWORD

This technical report presents results of research carried out by Dr. Moshe Carmeli of the Theoretical Physics Group of the General Physics Research Laboratory, Aerospace Research Laboratories, Project 7114. The research was initiated by the suggestion of Dr. Kenneth E. Kissell, Director of the General Physics Research Laboratory, in support of the Optical Properties of Space Objects (OPOS) program.

ABSTRACT

In this report an exact expression to the solar radiation pressure on a prolate spheroid is given.

INTRODUCTION

Recent studies by Fea¹, and by Fea and Smith², have shown the existence of an unexplained acceleration on spacecrafts which are balloon satellites of large area to mass ratio. It was suggested that the unexplained acceleration might be caused by solar radiation pressure. There is evidence^{3,4} that one of the satellites showing this anomalous acceleration is no longer spherical, and that it is probably shaped like a prolate spheroid. Previously, it had been assumed that the solar radiation scattered by the satellite is symmetrical about the satellite-sun line, an assumption which no longer holds for a prolate spheroid. It is therefore expected that additional perturbations of the orbit will arise.

Consequently, Smith and Fea⁵ developed a perturbation method to calculate the radiation pressure on a prolate spheroid. To this end two major assumptions were made: (1) the effective direction of reflection of the specular flux is determined by Snell's law on the incident ray that passes through the center of the satellite; and (2) that the magnitude of the flux reflected in this direction approximates to that which would be reflected by a sphere of surface area equal to that of the spheroid. Both assumptions seem not be valid in general.

In this paper we give an exact expression to the radiation pressure on a prolate spheroid without making any one of the two assumptions mentioned above.

PRELIMINARIES

It will be assumed that the solar radiation in the vicinity of the satellite is homogeneous. Also it will be assumed (see Introduction) that the satellite has the shape of a prolate spheroid. The relative orientation of the satellite with respect to the direction of light is then determined by the angle between the semi-major axis and direction of light.

We will use Cartesian coordinates defined, as usual, by $x = R \sin \theta \cos \varphi$, $y = R \sin \theta \sin \varphi$, and $z = R \cos \theta$. Let the prolate spheroid be located at the origin of coordinates and described by the equation

$$\frac{x^2 + y^2}{a^2} + \frac{z^2}{c^2} = 1. \quad (1)$$

Also, let the light rays be parallel to the x-z plane, having the direction given by (see Figure 1)

$$\vec{N} = (-\sin \theta_0, 0, -\cos \theta_0). \quad (2)$$

RADIATION PRESSURE

An infinitesimal light beam reflected from a surface element dS of the spheroid will apply a force given by

$$d\vec{F} = 2I (\vec{n} \cdot \vec{N}) d\vec{S}, \quad (3)$$

where I is the light density, \vec{n} is a unit vector perpendicular to the

surface element, and \vec{N} is the direction of light, which is given by Eq. (2). The geometry of the surface of the prolate spheroid determines both \vec{n} and $d\vec{S}$. They are given by⁶

$$\vec{n} = a g^{-\frac{1}{2}} \sin \theta (c \sin \theta \cos \varphi, c \sin \theta \sin \varphi, a \cos \theta), \quad (4)$$

$$d\vec{S} = g^{\frac{1}{2}} d\theta d\varphi \vec{n}, \quad (5)$$

where

$$g = a^2 \sin^2 \theta (a^2 \cos^2 \theta + c^2 \sin^2 \theta). \quad (6)$$

The result is

$$d\vec{F} = - \frac{2I a \sin \theta (c \sin \theta \sin \theta_0 \cos \varphi + a \cos \theta \cos \theta_0)}{(a^2 \cos^2 \theta + c^2 \sin^2 \theta)^{\frac{1}{2}}} \times$$

$$\times (c \sin \theta \cos \varphi, c \sin \theta \sin \varphi, a \cos \theta) d\theta d\varphi. \quad (7)$$

The total force \vec{F} is, accordingly,

$$\vec{F} = \int_{\sigma} d\vec{F}. \quad (8)$$

Here σ stands for the surface shaded by light on which the integration should be carried out.

BOUNDARY CONDITION

The boundary condition (integration limits) of the integral of force (8) is determined by the curve C which is the boundary of the shaded area σ . This curve is given by the equation

$$\vec{N} \cdot \vec{n}(\theta, \varphi) = 0. \quad (9)$$

Using Eqs. (2) and (4) for \vec{N} and \vec{n} one obtains explicitly for C the equation

$$c \sin \theta_0 \sin \theta \cos \varphi + a \cos \theta_0 \cos \theta = 0. \quad (10)$$

This is the equation of a curve given in terms of spherical coordinates, $\varphi = \varphi(\theta)$, or

$$\varphi = \pm \arccos \left(-\frac{a}{c} \cot \theta_0 \cot \theta \right). \quad (11)$$

Notice that the curve $\varphi = \varphi(\theta)$ is located in the plane

$$c \sin \theta_0 x + a \cos \theta_0 z = 0. \quad (12)$$

In Figure 2 we give the boundary limits where the integration on the angles θ and φ should be carried out.

TOTAL FORCE

The force integral (8) can now be given explicitly as

$$F_x = -2I ac \iint \frac{\sin^2 \theta (c \sin \theta_0 \sin \theta \cos \varphi + a \cos \theta_0 \cos \theta) \cos \varphi}{(a^2 \cos^2 \theta + c^2 \sin^2 \theta)^{\frac{3}{2}}} d\varphi d\theta \quad (13a)$$

$$F_y = -2I ac \iint \frac{\sin^2 \theta (c \sin \theta_0 \sin \theta \cos \varphi + a \cos \theta_0 \cos \theta) \sin \varphi}{(a^2 \cos^2 \theta + c^2 \sin^2 \theta)^{\frac{3}{2}}} d\varphi d\theta \quad (13b)$$

$$F_z = -2I a^2 \iint \frac{\sin \theta \cos \theta (c \sin \theta_0 \sin \theta \cos \varphi + a \cos \theta_0 \cos \theta)}{(a^2 \cos^2 \theta + c^2 \sin^2 \theta)^{\frac{3}{2}}} d\varphi d\theta \quad (13c)$$

As can be seen from Figure 2. the limits of the double integrals in Eqs. (13) should be taken from $\varphi = -\pi$ to $\varphi = +\pi$ for $0 \leq \theta \leq \theta_1$, and from $\varphi = -\arccos(-\frac{a}{c} \cot \theta_0 \cot \theta)$ to $\varphi = +\arccos(-\frac{a}{c} \cot \theta_0 \cot \theta)$ for $\theta_1 \leq \theta \leq \pi - \theta_1$. The angle θ_1 is that θ for which the angle φ of the curve C satisfies $\varphi(\theta_1) = \pi$.

The force term F_y , Eq. (13b), can easily be shown to be equal to zero.

CASE OF A SPHERE

The force terms given by (13) have a particularly simple form in the case of a sphere (i.e. $a = c$). One obtains in this case

$$F_x^S = -2I a^2 \iint \sin^2 \theta (\sin \theta_0 \sin \theta \cos \varphi + \cos \theta_0 \cos \theta) \cos \varphi d\varphi d\theta \quad (14a)$$

$$F_y^S = 0 \quad (14b)$$

$$F_z^S = -2I a^2 \iint \sin \theta \cos \theta (\sin \theta_0 \sin \theta \cos \varphi + \cos \theta_0 \cos \theta) d\varphi d\theta \quad (14c)$$

One can simplify these integrals by going into new variables θ' , φ' in which the integration limits are not dependents on each other. To this end one rotates the coordinate system around the y-axis:

$$\begin{aligned} x &= x' \cos \theta_0 + z' \sin \theta_0 \\ z &= -x' \sin \theta_0 + z' \cos \theta_0 \end{aligned} \quad (15)$$

A simple calculation, using the relations $x = a \sin \theta \cos \varphi$, $y = a \sin \theta$

$\sin \varphi$, $z = a \cos \theta$ and $x' = a \sin \theta' \cos \varphi'$, $y' = a \sin \theta' \sin \varphi'$,
 $z' = a \cos \theta'$, then shows that θ and φ are related to the new variables
 θ' and φ' by

$$\begin{aligned}\sin \theta' \cos \varphi' \cos \theta_0 + \cos \theta' \sin \theta_0 &= \sin \theta \cos \varphi, \\ -\sin \theta' \cos \varphi' \sin \theta_0 + \cos \theta' \cos \theta_0 &= \cos \theta,\end{aligned}\quad (16)$$

and

$$\sin \theta' d\theta' d\varphi' = \sin \theta d\theta d\varphi. \quad (17)$$

As a result one obtains for the force components acting on the sphere

$$\begin{aligned}F_x^S &= -2I a^2 \int_{\theta=0}^{\pi/2} \int_{\varphi=-\pi}^{+\pi} (\sin \theta' \cos \varphi' \cos \theta_0 + \cos \theta' \sin \theta_0) \cos \theta' \\ &\quad \sin \theta' d\varphi' d\theta' \\ F_y^S &= 0 \\ F_z^S &= -2I a^2 \int_{\theta=0}^{\pi/2} \int_{\varphi=-\pi}^{+\pi} (-\sin \theta' \cos \varphi' \sin \theta_0 + \cos \theta' \cos \theta_0) \cos \theta' \\ &\quad \sin \theta' d\varphi' d\theta' \\ &\quad (18)\end{aligned}$$

The result is

$$\begin{aligned}F_x^S &= -(4/3)\pi a^2 I \sin \theta_0 \\ F_y^S &= 0 \\ F_z^S &= -(4/3)\pi a^2 I \cos \theta_0\end{aligned}\quad (19)$$

REFERENCES

1. Fea, K., "The Orbital Acceleration of High Balloon Satellites", presented at the XII Plenary Meeting of COSPAR, Prague, May 1969.
2. Fea, K. and Smith, D. E., "Some Further Studies of Perturbation of Satellites at Great Altitude", Planetary and Space Science, to be published.
3. Vanderburgh, R. C., "Wideband (Visual Spectrum) Photoelectric Photometry of PAGEOS During Its First Fifteen Months in Orbit", in Proceedings of the University of Miami Symposium on Optical Properties of Orbiting Satellites, D. Duke and K. E. Kissell, Editors, May 1969.
4. Vanderburgh, R. C., "ARL OPOS Photoelectric photometry of PAGEOS on 4 July 1968", ARL-MR-68-0004, August 1968.
5. Fea, K. and Smith, D. E., "Radiation Pressure Effects on the Acceleration of High Altitude Balloon Satellites", Goddard Space Flight Center Technical Report X-550-70-92, Greenbelt, Maryland, March 1970.
6. Tamburino, L., private communication.

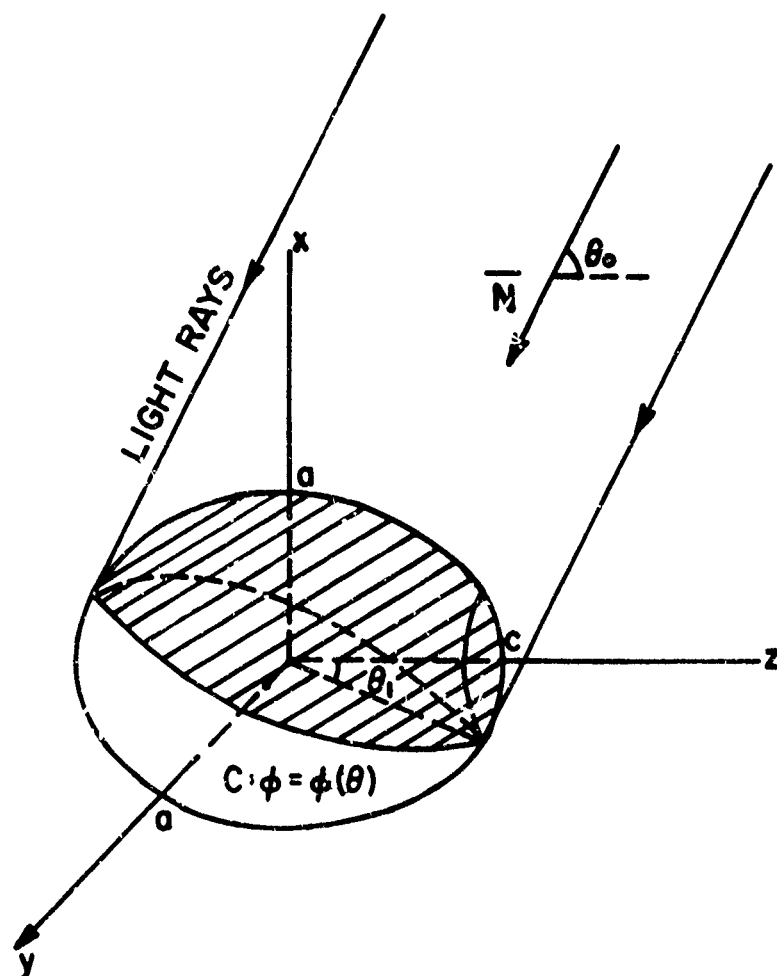


Figure 1. Prolate ellipsoid and light rays in the x-z plane along the vector $\vec{N} = (-\sin \theta_0, 0, -\cos \theta_0)$. The curve C is defined by $\vec{N} \cdot \vec{n}(\theta, \phi) = 0$, where the vector \vec{n} is normal to the surface of the ellipsoid.

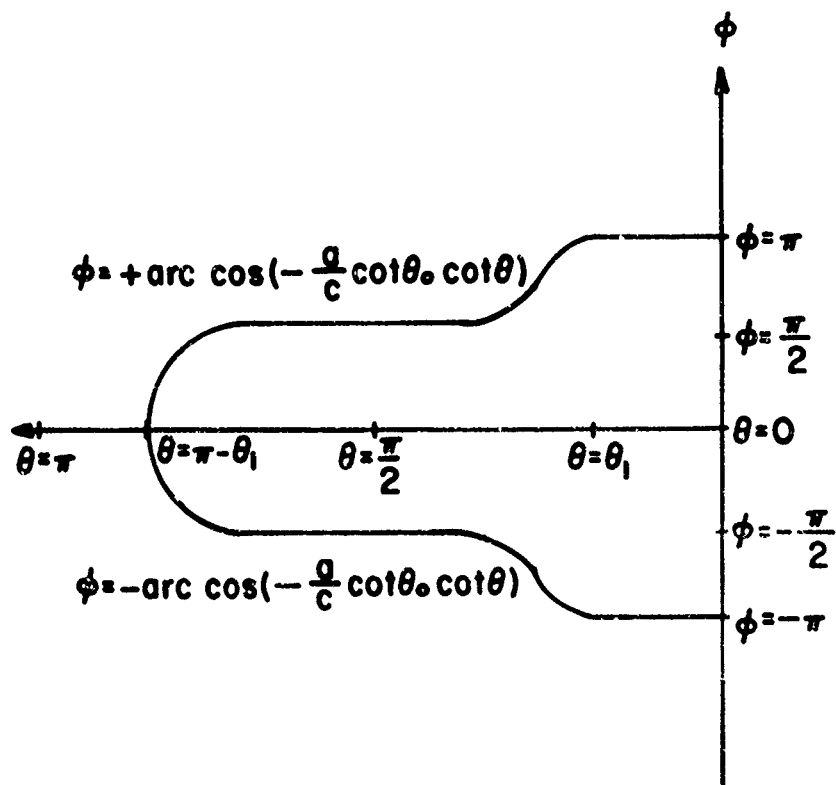


Figure 2. Boundary limit diagram. The boundary of integration is given by $\varphi = \pm \pi$ for $0 \leq \theta \leq \theta_1$, and $\varphi(\theta) = \pm \arccos(-\frac{a}{c} \cot \theta_0 \cot \theta)$ for $\theta_1 \leq \theta \leq \pi - \theta_1$. The angle θ_1 is defined by $\theta_1 = \arctg(\frac{a}{c} \cot \theta_0)$.

Security Classification

DOCUMENT CONTROL DATA - R & D

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

1. ORIGINATING ACTIVITY (Corporate author) General Physics Research Laboratory Aerospace Research Laboratories Wright-Patterson AFB, Ohio 45433		2a. REPORT SECURITY CLASSIFICATION UNCLASSIFIED	
		2b. GROUP	
3. REPORT TITLE "Solar Radiation Pressure on High Altitude Satellite"			
4. DESCRIPTIVE NOTES (Type of report and inclusive dates) Technical Report			
5. AUTHOR(S) (First name, middle initial, last name) MOSHE CARMELI			
6. REPORT DATE February 1972		7a. TOTAL NO OF PAGES 13	7b. NO OF REFS 6
8a. Source In-House Research		9a. ORIGINATOR'S REPORT NUMBER(S)	
b. PROJECT NO 7114			
c. DOD Element - 61102F		9b. OTHER REPORT NO(S) (If other numbers that may be assigned this report)	
d. DOD Subelement - 681301		ARL 72-0034	
10. DISTRIBUTION STATEMENT Approved for public release; distribution unlimited.			
11. SUPPLEMENTARY NOTES TECH OTHER		12. SPONSORING MILITARY ACTIVITY ARL/LG WPAFB, Ohio	
13. ABSTRACT In this report an exact expression to the solar radiation pressure on a prolate spheroid is given.			

DD FORM 1 NOV 4 1473

Security Classification

Security Classification

14 KEY WORDS	LINK A		LINK B		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT